











Experimental setup

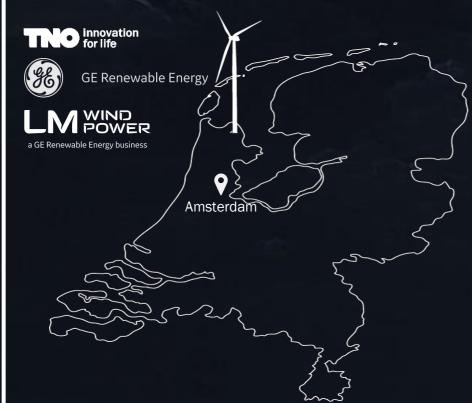
Numerical simulations

Results

Takeaways

EXPERIMENTAL SETUPPROJECT OVERVIEW

TIADE (Turbine Improvements for Additional Energy)



Scope of TIADE project:

- > Blade improvement (innovative tip shapes, VGs, turbulator)
- Validation (erosion, yawed inflow, stall and/or vortex induced vibrations)
- Measurement innovations (aerodynamic pressure, torsion deformation, fibre optics)

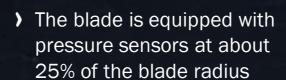
Scope of PhD:

- Development of an aeroelastically tailored tip
- Numerical aeroelastic simulations & comparison to field data



EXPERIMENTAL SETUPDATA ACQUISITION

) A LiDAR system is used to measure the wind field

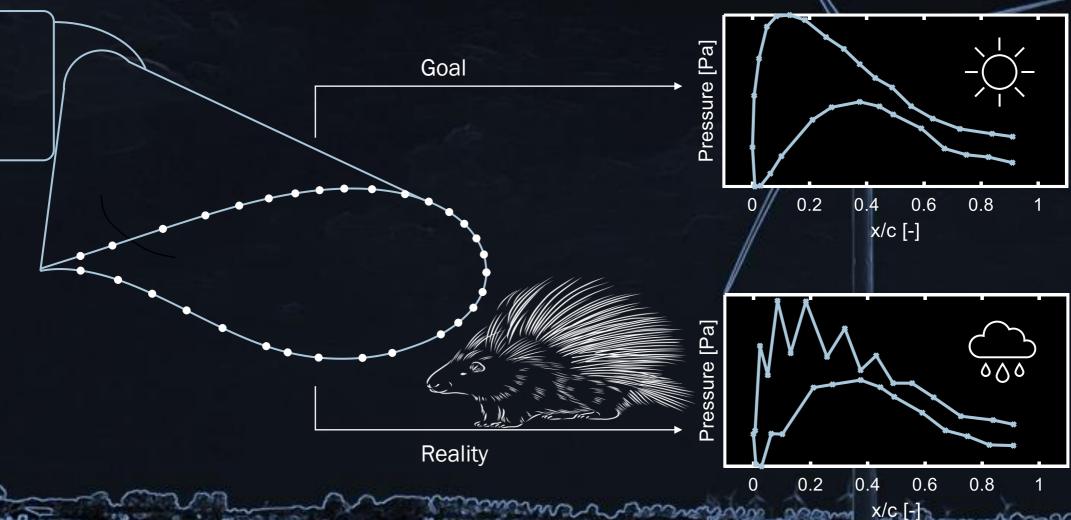






-) Operational conditions are monitored
 -) Power
 -) Rotational speed
 -) Rotor azimuth
 -) Blade pitch
 - **)** ..

EXPERIMENTAL SETUP CHALLENGES





NUMERICAL SIMULATIONS SIMULATION TOOLS (1/2)

Input:

- Aerodynamic and structural turbine model
-) Operating conditions

Output:

- **)** Power, thrust
- **)** Blade aerodynamics, moments and deformations

Input:

- Airfoil geometry
- > Reynolds number, Angle of attack, c/r value

Output:

- **)** Airfoil polars: c_d , c_l , c_m
- **>** Pressure coefficient distribution



) Phatas

-) BEM
- Structural solver



-) 2D panel method
-) Based on XFOIL
- Modified for rotating airfoils



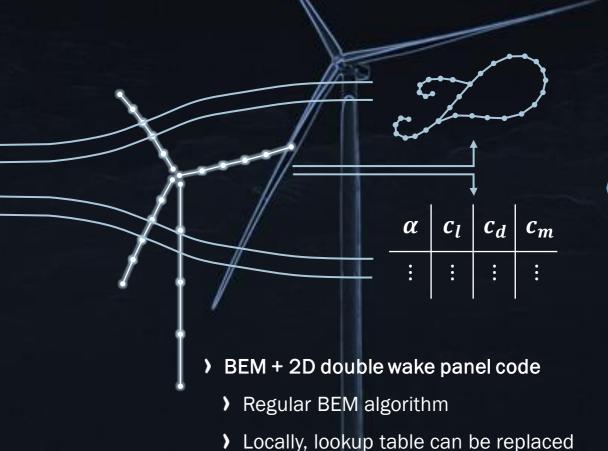
NUMERICAL SIMULATIONS SIMULATION TOOLS (2/2)

Input:

- Aerodynamic and structural turbine model
- Airfoil geometry
-) Operating conditions

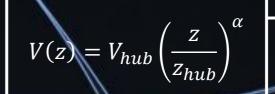
Output:

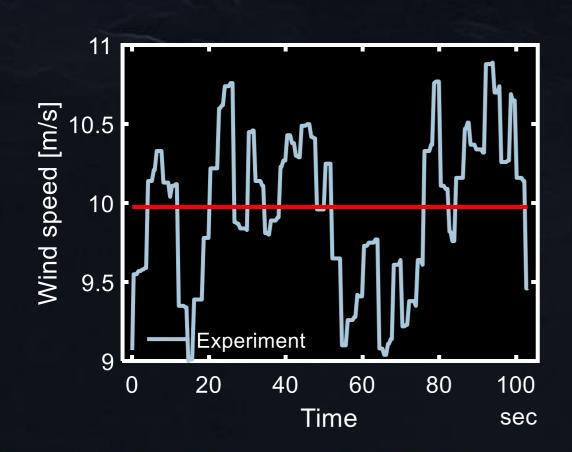
- **)** Power, thrust
- **)** Blade aerodynamics, moments and deformations
-) Unsteady pressure distributions

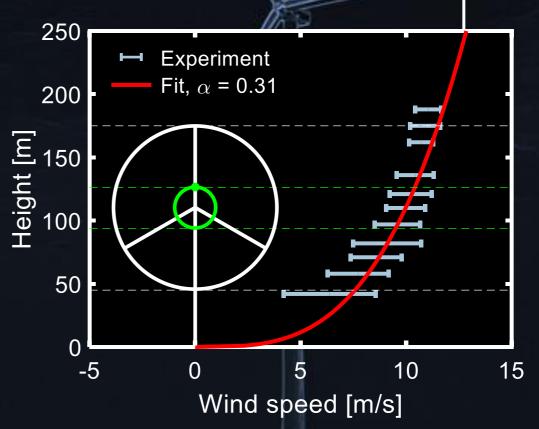


by 2d double wake panel method

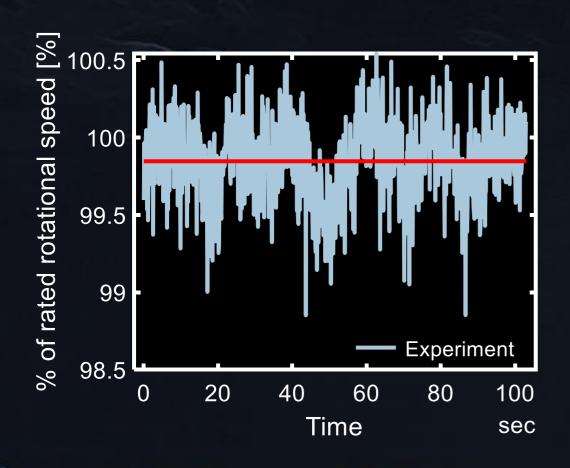
RESULTS OPERATING CONDITIONS (1/2)

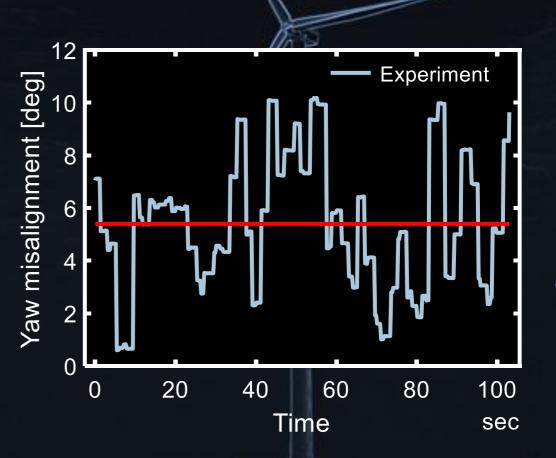




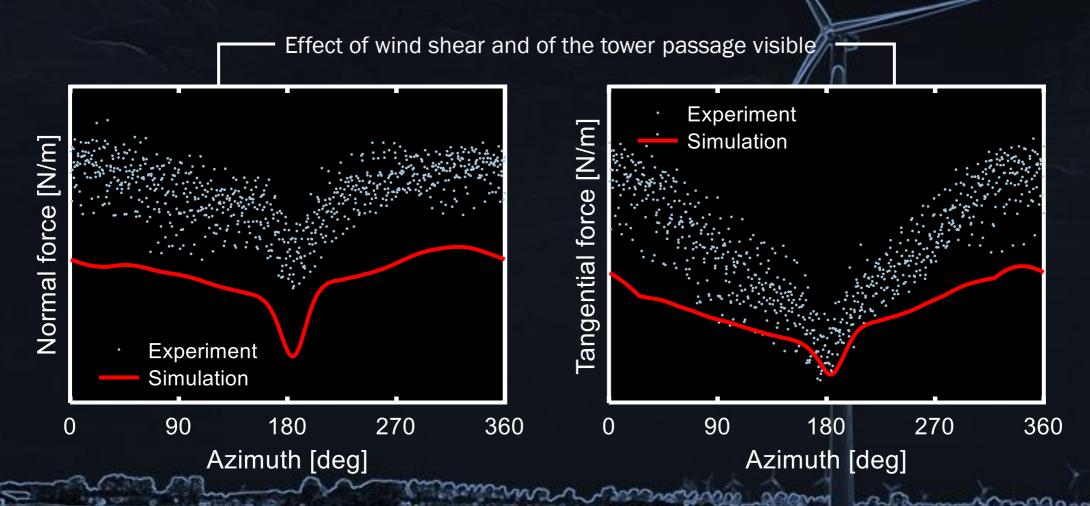


RESULTSOPERATING CONDITIONS (2/2)

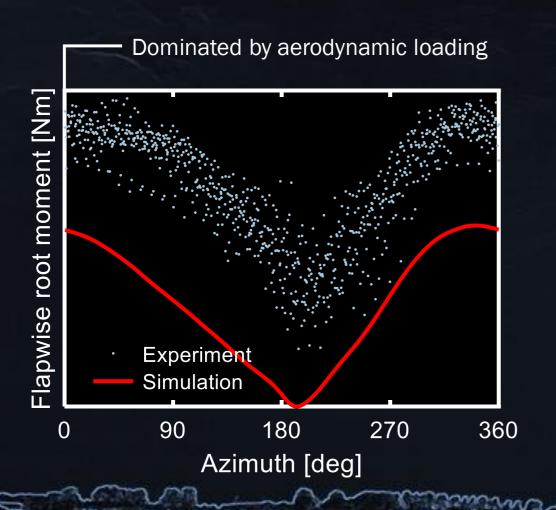


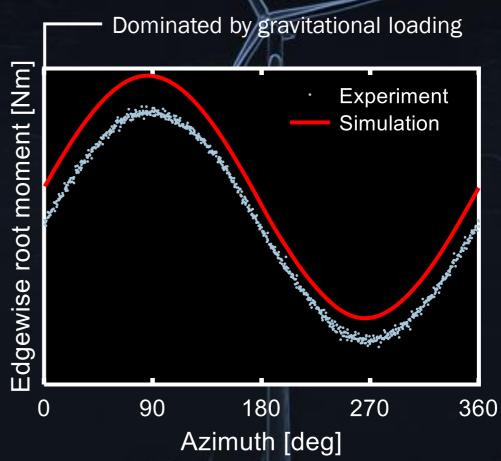


RESULTSSECTIONAL FORCES



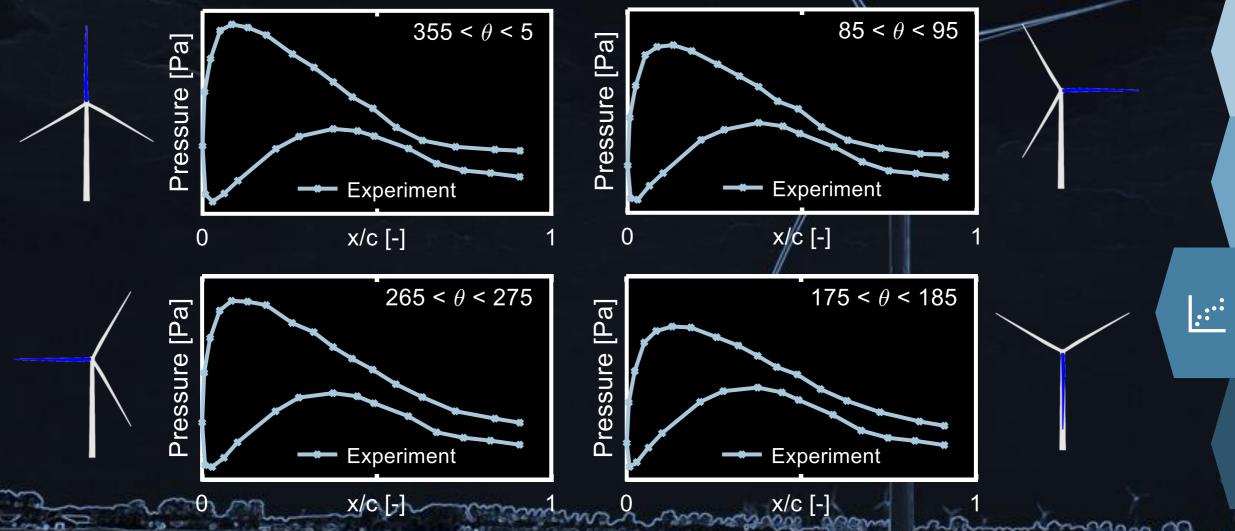
RESULTSBLADE ROOT BENDING MOMENTS



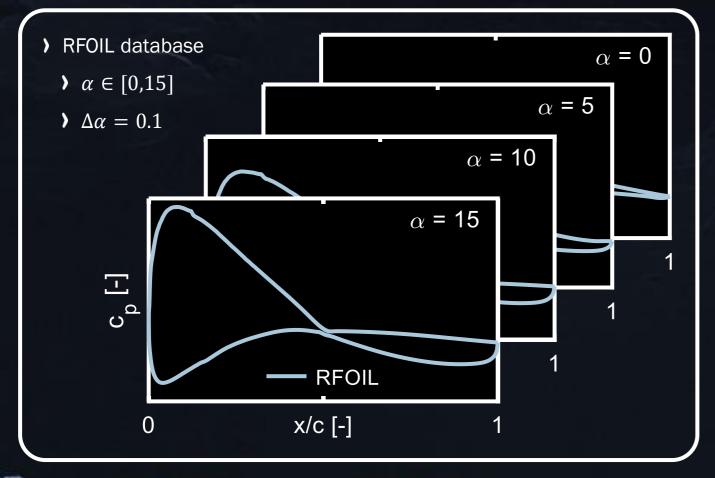


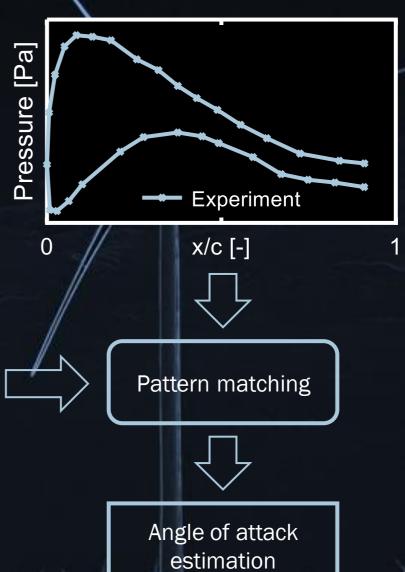
RESULTS

PRESSURE VARIATION DURING ONE ROTATION



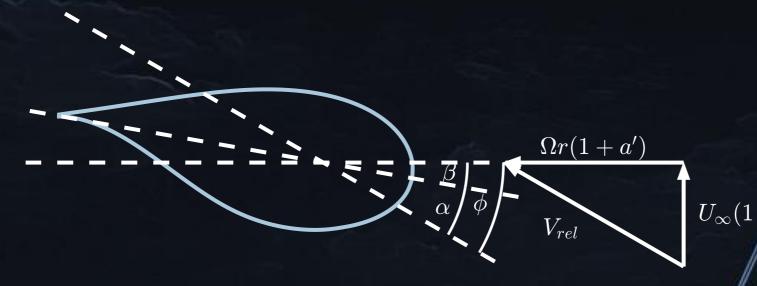
RESULTSANGLE OF ATTACK ESTIMATION





RESULTS

RELATIVE VELOCITY ESTIMATION



Pitch and twist angle	β	√
Inflow angle	φ	√
Rotational velocity	Ωr	√
Free stream velocity	U_{∞}	√
Axial induction factor	а	√
Tangential induction factor	a'	

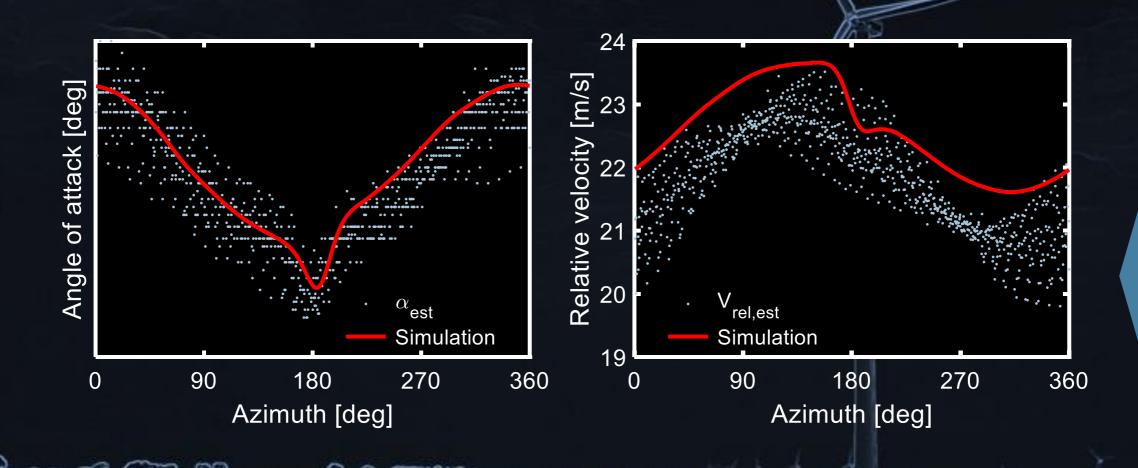
Angle of attack

Relative velocity

- $\phi = \alpha + \beta$
- $U_{\infty}(1-a) = \Omega r(1+a') \tan \phi \approx \Omega r \tan \phi$
- $V_{rel} = \sqrt{\left(U_{\infty}(1-a)\right)^2 + (\Omega r)^2}$

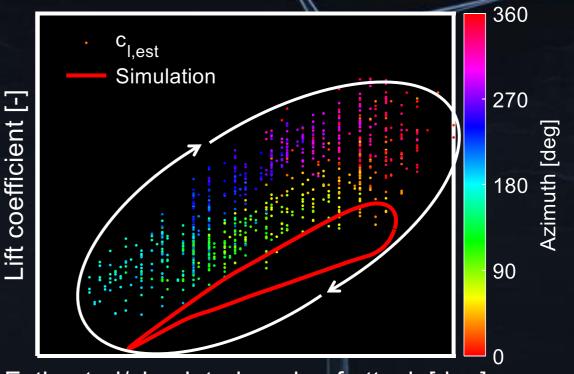
 V_{rel} \checkmark

RESULTSANGLE OF ATTACK AND RELATIVE VELOCITY



RESULTSUNSTEADY LIFT COEFFICIENT CURVE

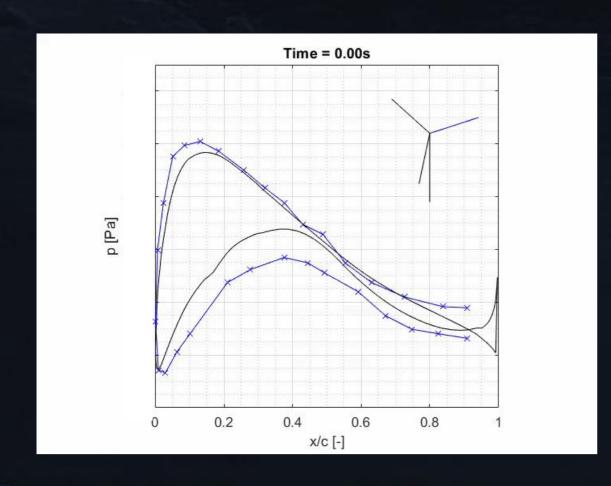
- Simulations use Snel's dynamic stall model of 1st order
-) Hysteresis effect clearly visible in simulation and experiment
-) Magnitude of $c_l = \frac{L}{\frac{\rho}{2}V_{rel}^2c}$ influenced by
 -) Lift force (simulations underpredict experiment)
 - Relative velocity (simulations overpredict experiment)



Estimated/simulated angle of attack [deg]

RESULTS

DOUBLE WAKE CODE VS MEASURED PRESSURE



Animation shows:

- Turbine rotation, blade equipped with sensors in blue
- Measured pressure distribution (blue)
- Simulated pressure distribution (black)

Findings:

- Detailed insight into sectional aerodynamics
- Partly already good agreement between experiment and simulation
- Overprediction of the suction peak due to inviscid formulation of double wake code

AEROELASTIC WIND TURBINE BLADE BEHAVIOUR TAKEAWAYS

- > Field experiments bring a lot of challenges in data acquisition
- **)** Experimental data is very valuable for the validation of numeric models
-) Good agreement between simulations and experiment regarding shape and range but not in magnitude
- Combined BEM + double wake panel code offers detailed insight into airfoil aerodynamics
- More engineering work required to make up for missing physics and improve code accuracy



THANK YOU FOR YOUR TIME

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GE Renewable Energy





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